

# Thermomechanical Design Of A Gas Turbine Reheat Combustor Experiment Using Finite Element Analysis

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## Introduction and Outline

- Sequential combustion - framework for operationally flexible and low-emission gas turbines [1]
- Experimental research on lean premixed, auto-ignition stabilized flame dynamics
- **Thermomechanical design optimization** of a reheat combustor experiment

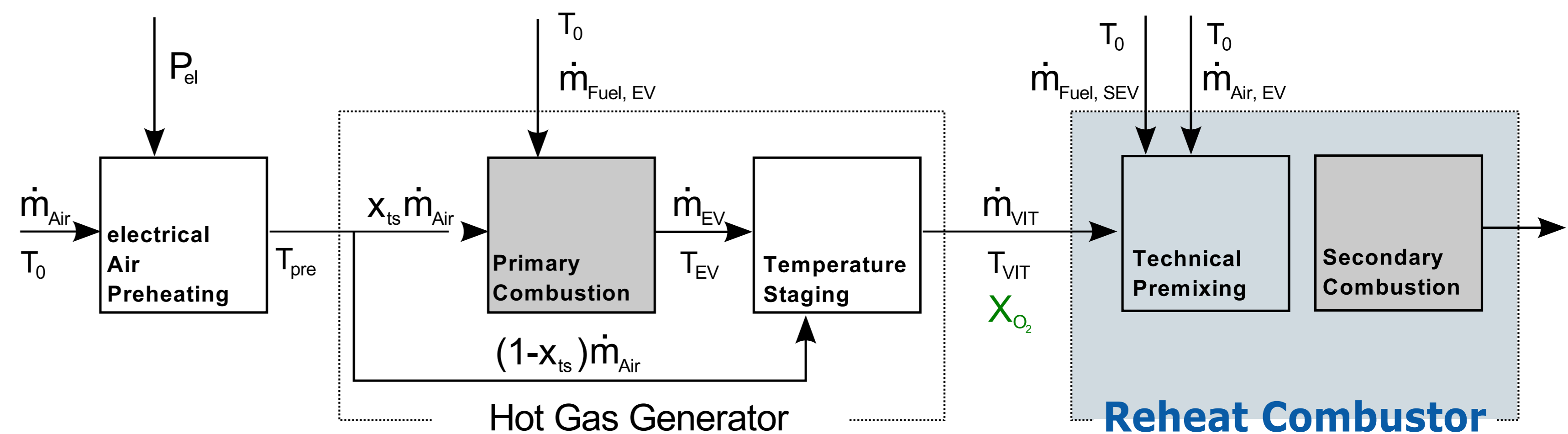


Figure 1. Schematic of the reheat combustor experiment

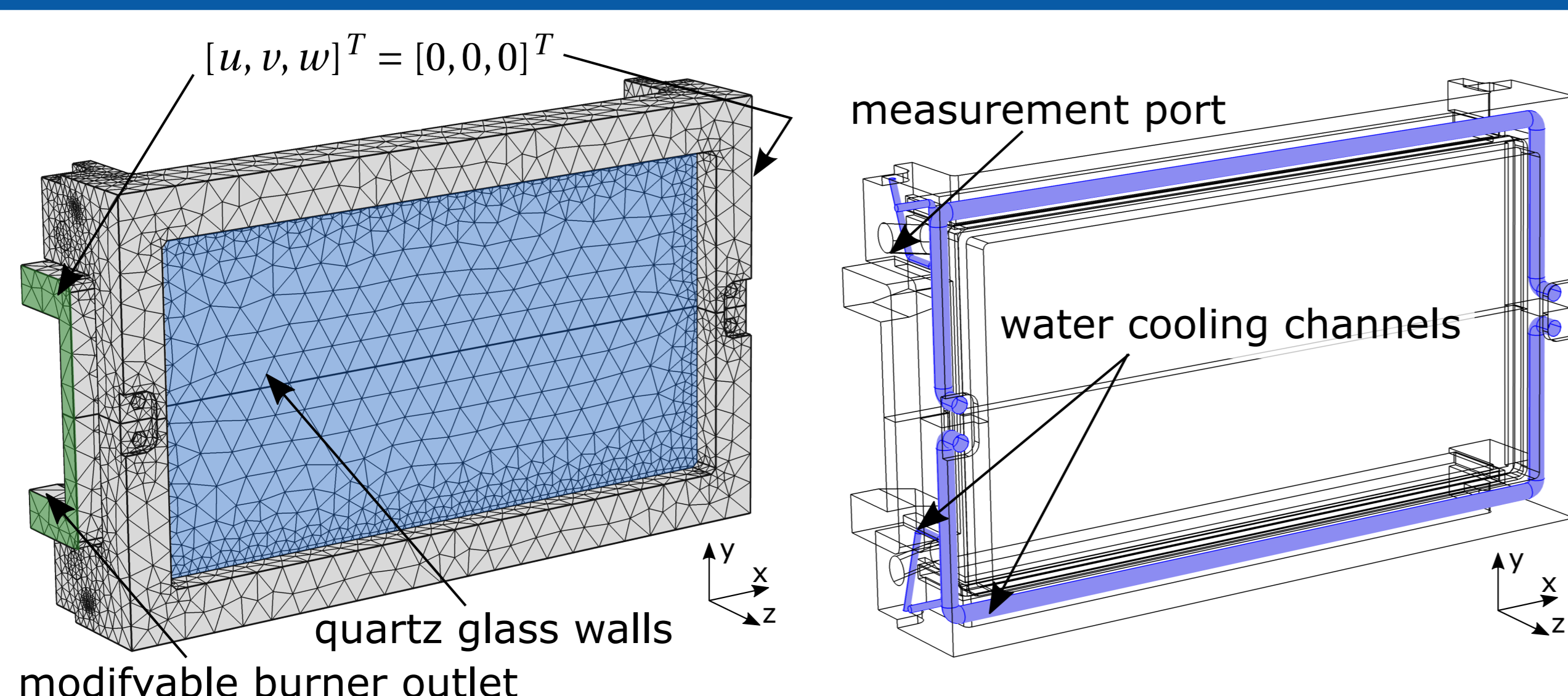


Figure 2. Combustion chamber geometry and FE analysis conditions

## Thermomechanical Design/Requirements

- High thermal loads - 20 MW/m<sup>2</sup>bar, 1300-2200 K
- Application of high fidelity **conventional** and **optical diagnostic techniques**
- **Material coupling** with different thermal expansion - air-tight design of steel and quartz glass structure
- **Marginal heat loss** to sustain auto-ignition limits

## Thermomechanical Finite Elemente Analysis

### 1. Thermal Analysis

- ▶ Heat conduction solution  $\nabla(k\nabla T(\mathbf{x})) = 0$ , in  $\Omega$
- ▶ Dirichlet boundaries from CHT simulations

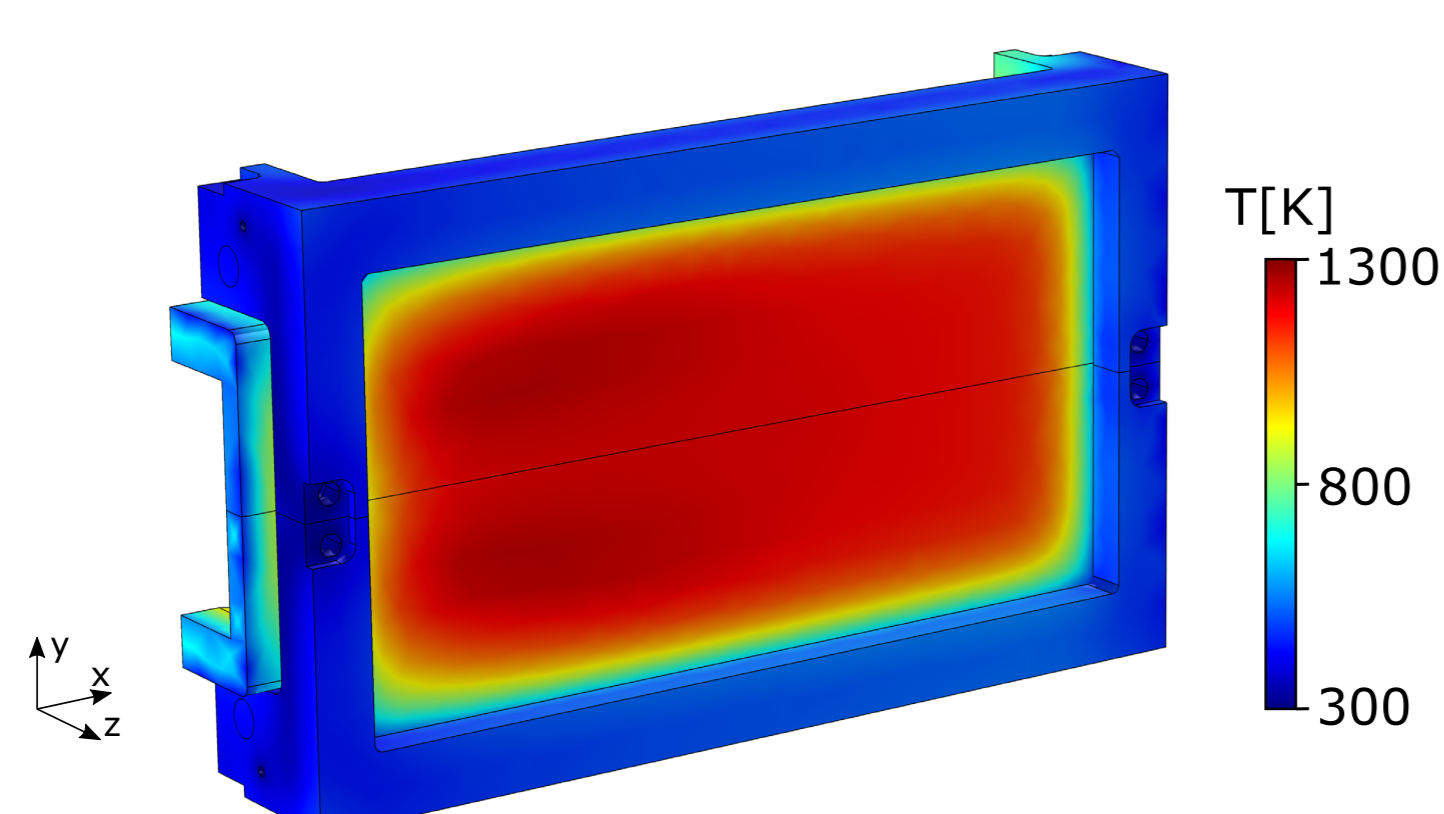


Figure 3. Temperature distribution

### 2. Structural Analysis

- ▶ Linear elastic, uncoupled, quasi-static thermomechanical solution [2,3]

$$\nabla\sigma = 0, \text{ on } \Omega \quad \text{Equilibrium equation}$$

$$\sigma = D(\varepsilon - \varepsilon_0) \quad \text{constitutive equations}$$

$$\varepsilon = \frac{1}{2} [\nabla\mathbf{u} + (\nabla\mathbf{u})^T] \quad \text{strain-displacement relations}$$

### COMSOL Multiphysics

- Implementation of weak form PDE formulation
- User-defined Interpolation Function to ascribe locally varying material properties

spatial coordinates			material properties			
x	y	z	$k[\frac{W}{m\cdot K}]$	$E[\text{Pa}]$	$\nu[-]$	$\alpha[K^{-1}]$

Figure 4. Material properties IF

## Results

- ▶ Relative displacement and structural deformation

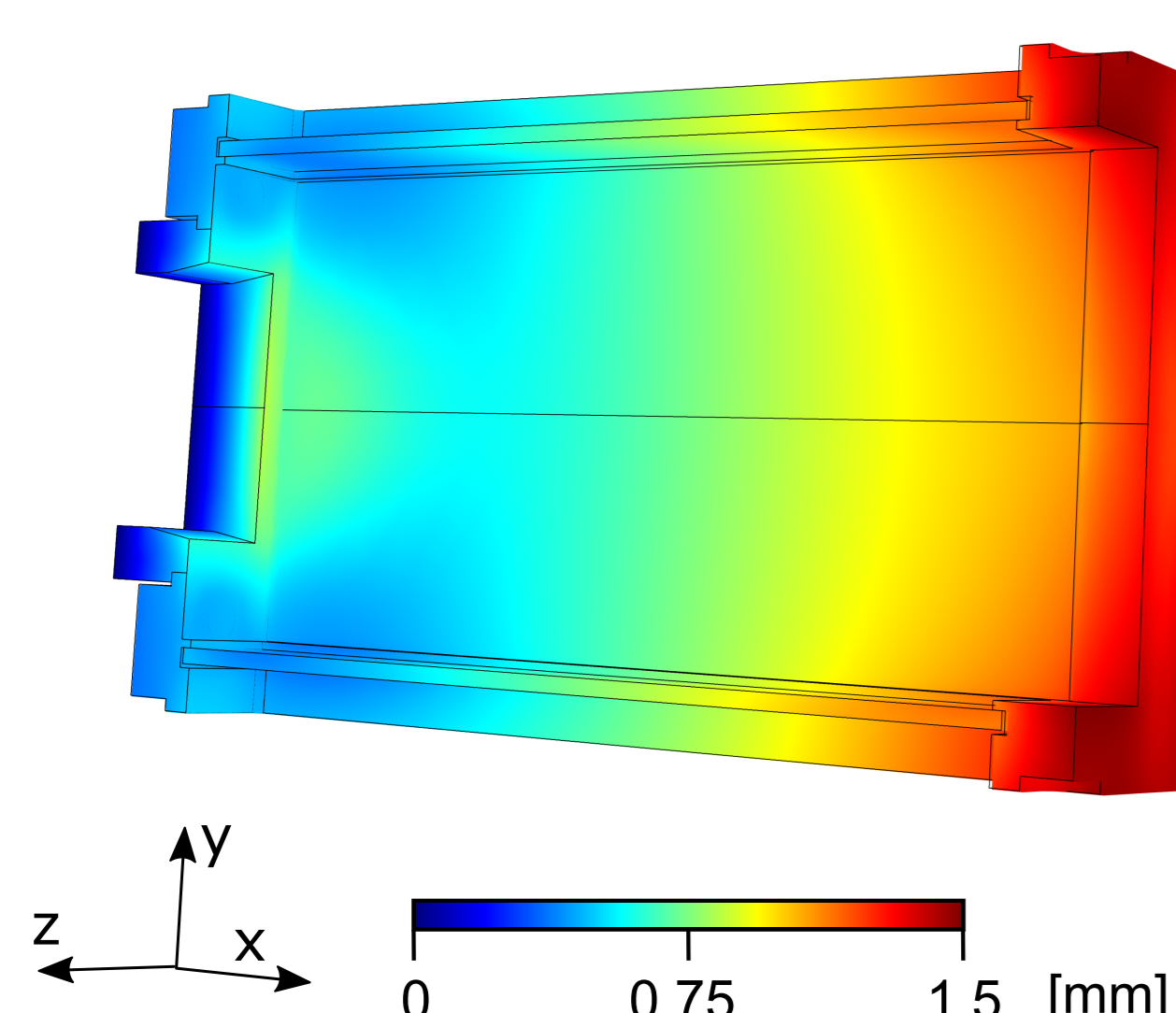


Figure 5. Global total displacement field

- ▶ Localization of critical thermally induced stresses

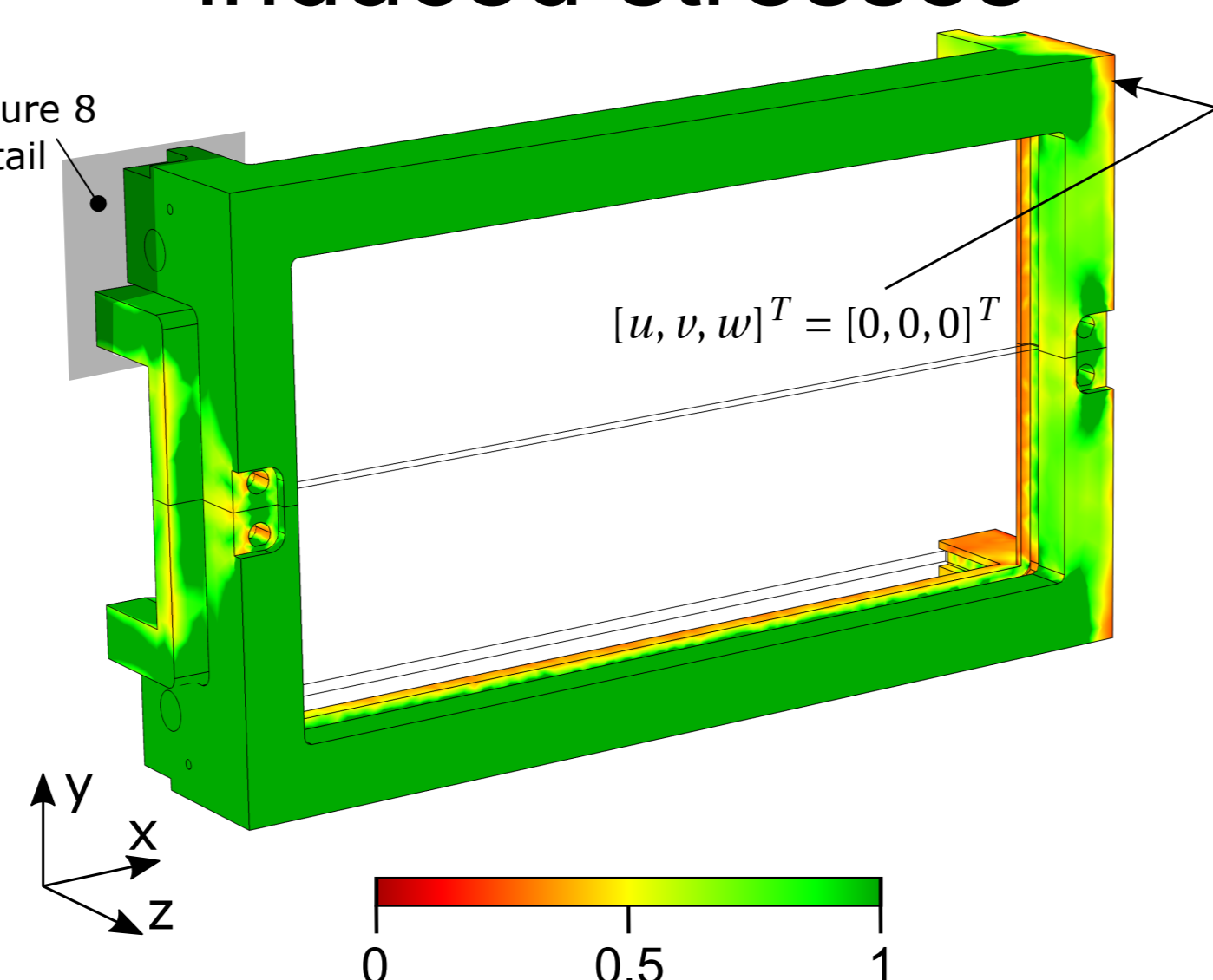


Figure 7. Linear elasticity stress ratio:  $R_{p0.2}/\sigma_{Mises}$

- ▶ Cooling optimization for instrumentation ports

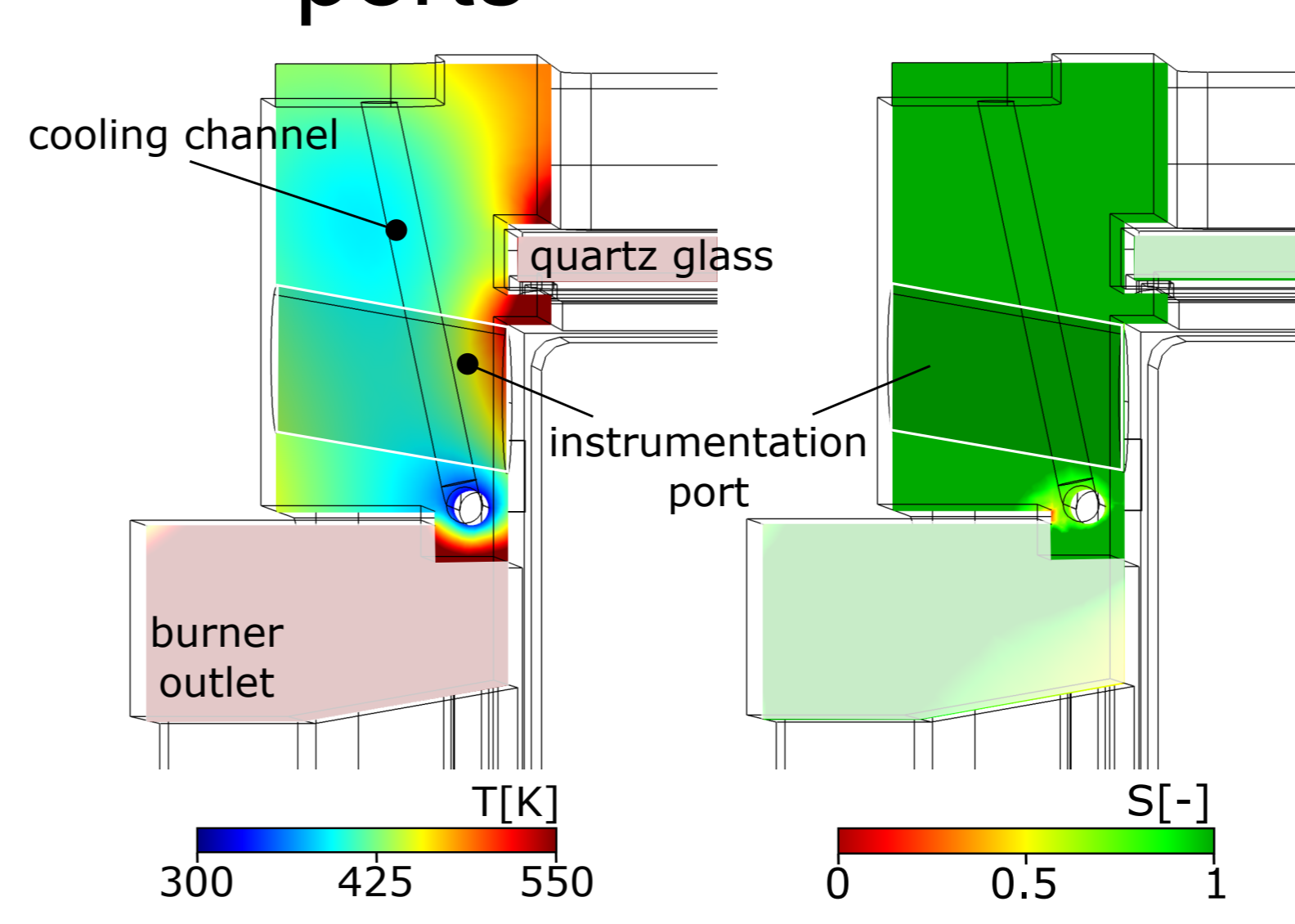


Figure 6. Instrumentation port detail

## References

- [1] Joos, F., Brunner, P., Schulte-Werning, B., Syed, K., and Eroglu, A., "Development of the Sequential Combustion System for the ABB GT24/26 Gas Turbine Family", ASME Paper No 1996-GT-315, (1996)
- [2] Zienkiewicz, O., Taylor, R., Zhu, J., "The Finite Element Method - Its Basics and Fundamentals", Butterworth-Heinemann, (2013)
- [3] Boley, B., Weiner, J., "Theory of Thermal Stresses", General Publishing Company Ltd., (1997)

